

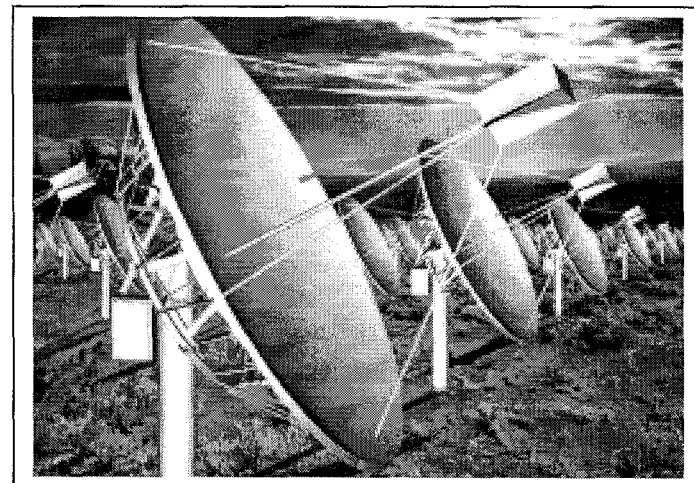


Large Antenna Arrays for the DSN and Radio Astronomy

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August 9, 2001

- The Existing DSN and Future Arrays
- What a Square-Km Array (SKA) Could do for DSN
- Technology Development for Large Arrays
- Costs of Large Arrays
- Situation Summary
- Next Steps for JPL



Comparison of Existing Large Antennas and Future Arrays

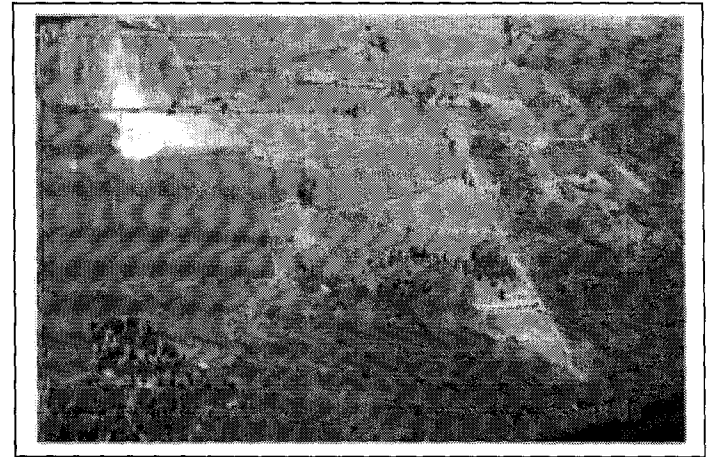
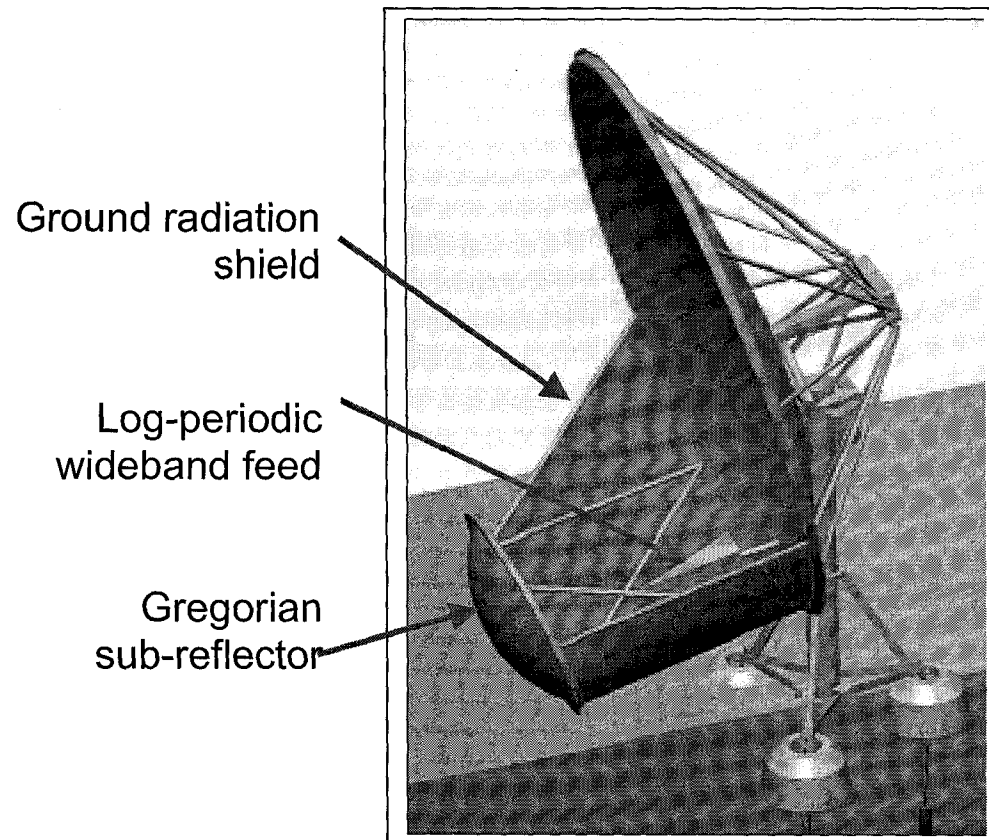
Antenna	Elements	Effective Area	Upper Frequency	Tsys	A/Tsys	Year Finished
DSN 70m	1 x 70 m	2,607	8 GHz	18	145	1965
GBT	1 x 100 m	5,700	100 GHz	20	285	2000
VLA	27 x 25 m	8,978	43 GHz	32	280	1982
Arecibo	1 x 305 m	23,750	8 GHz	25	950	1970
ALMA	64 x 12 m	4,608	800 GHz	50	92	2011
ATA	350 x 6 m	6,703	11 GHz	35	192	2005
DSN Array	1600 x 8m	51,200	32 GHz	25	2048	2010
SKA	14,000 x 10m	700,000	32 GHz	35	20,000	2015

SKA Communication Improvement Relative to DSN 70m Antennas

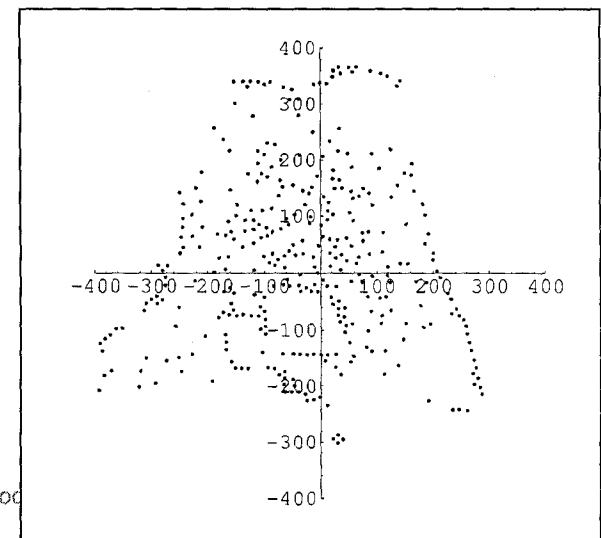
Antenna	dB Improvement at 8.4 GHz	dB Improvement at 32 GHz
DSN 70m	0	+6 dB
DSN 34m	-6 dB	0
SKA	21.3 dB	27.3 dB

What is the Allen Telescope Array, ATA?

- 350 x 6m offset hydroformed paraboloids
- Instantaneous bandwidth of 0.5 to 11 GHz
- A joint project of the SETI Institute and UC Berkeley
- Privately funded to cost approximately \$20M
- First array testing in 2003; completion in 2005



Hat Creek, CA site (above) and
array configuration (below)

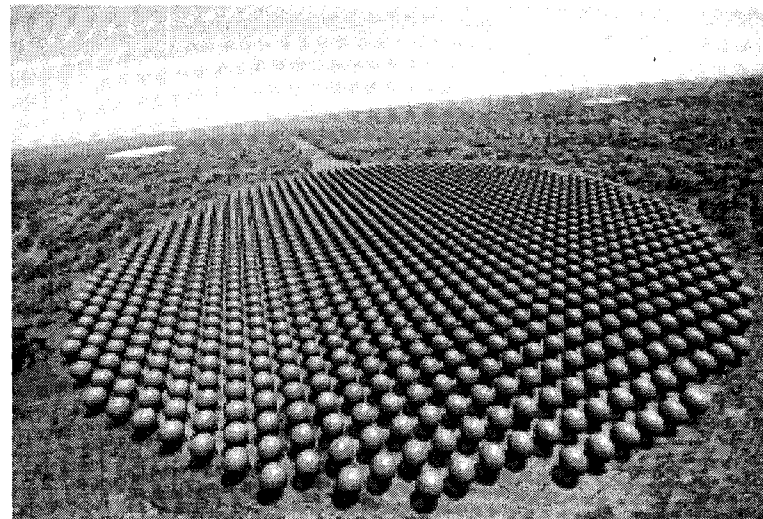


What is the SKA?

- An international project to design a very large area array for radio astronomy in the cm wavelength range.
- The web site, <http://www.skatelescope.com>, contains science justification and links to activities in several countries
- US approach is a large array ($\approx 10,000$) of small ($\approx 10\text{m}$) antennas, organized into a 1000km diameter spiral of ≈ 100 close packed stations
- Next scientific and technical meeting, *The SKA: Defining the Future*, Berkeley, CA, July 9-12. See web site for program and registration.
- Chairman of the International SKA Project, Ron Ekers, will meet with JPL Director, Charles Elachi, and give a **talk at JPL, August 16, 10am, in Bldg 167 conference room.**

Key Specifications

- $A_{\text{eff}}/T_{\text{sys}} > 20,000 \text{ m}^2/\text{K}$
(1 square km with $T_{\text{sys}}=50\text{K}$)
- Frequency, 0.15 – 20 GHz
- Resolution 35 nano-radians
(5km beam at 1 A.U. at 20GHz)



A Road Map for the United States' Development Efforts on the Square Kilometer Array

PREPARED BY THE US SKA CONSORTIUM

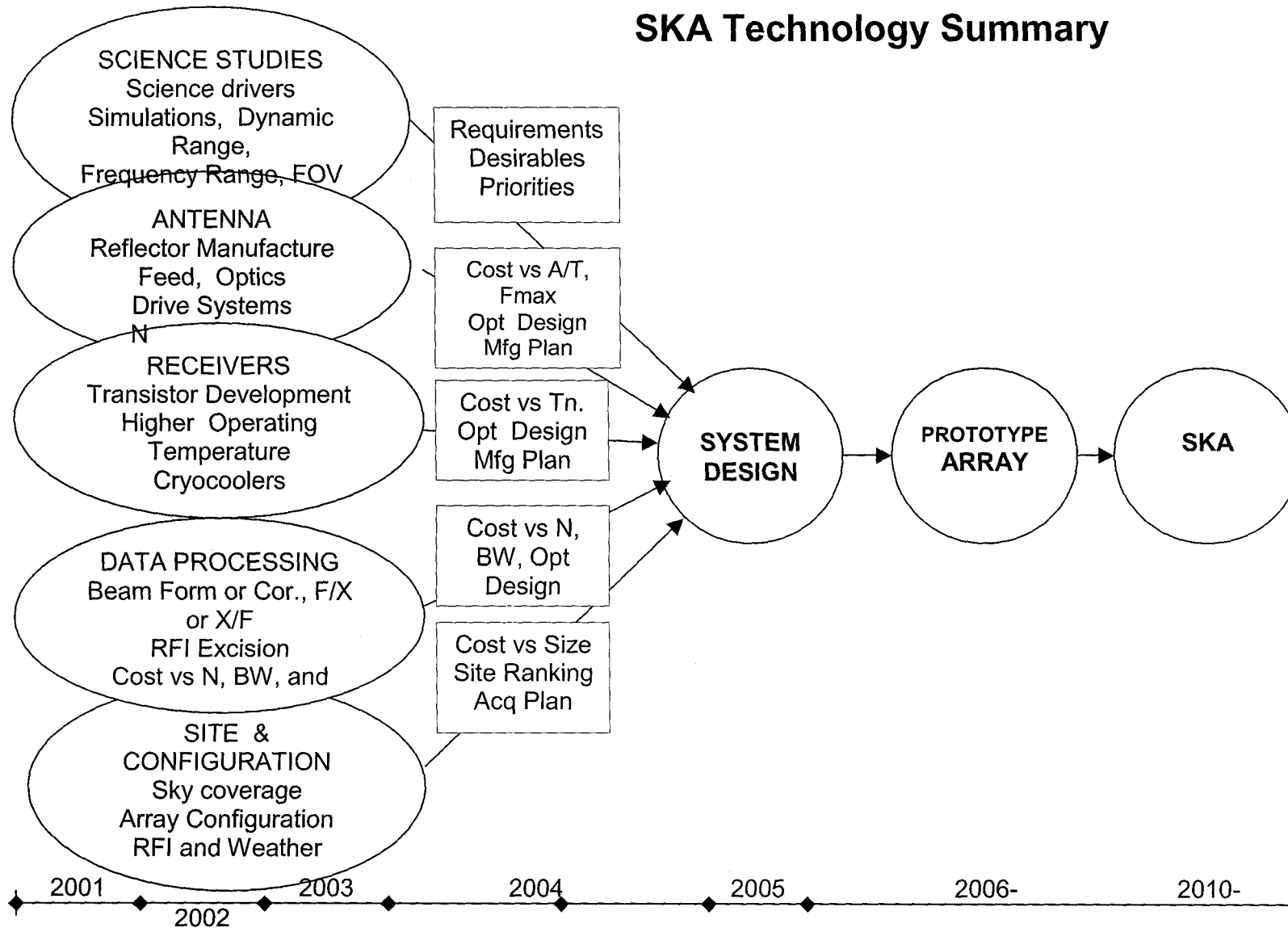
California Institute of Technology (including JPL)
Cornell University (including NAIC)
Harvard-Smithsonian Center for Astrophysics
Massachusetts Institute of Technology (including Haystack Observatory)
National Radio Astronomy Observatory
Naval Research Laboratory
Ohio State University
SETI Institute
University of California, Berkeley
University of Minnesota

Draft 2: March 28, 2001

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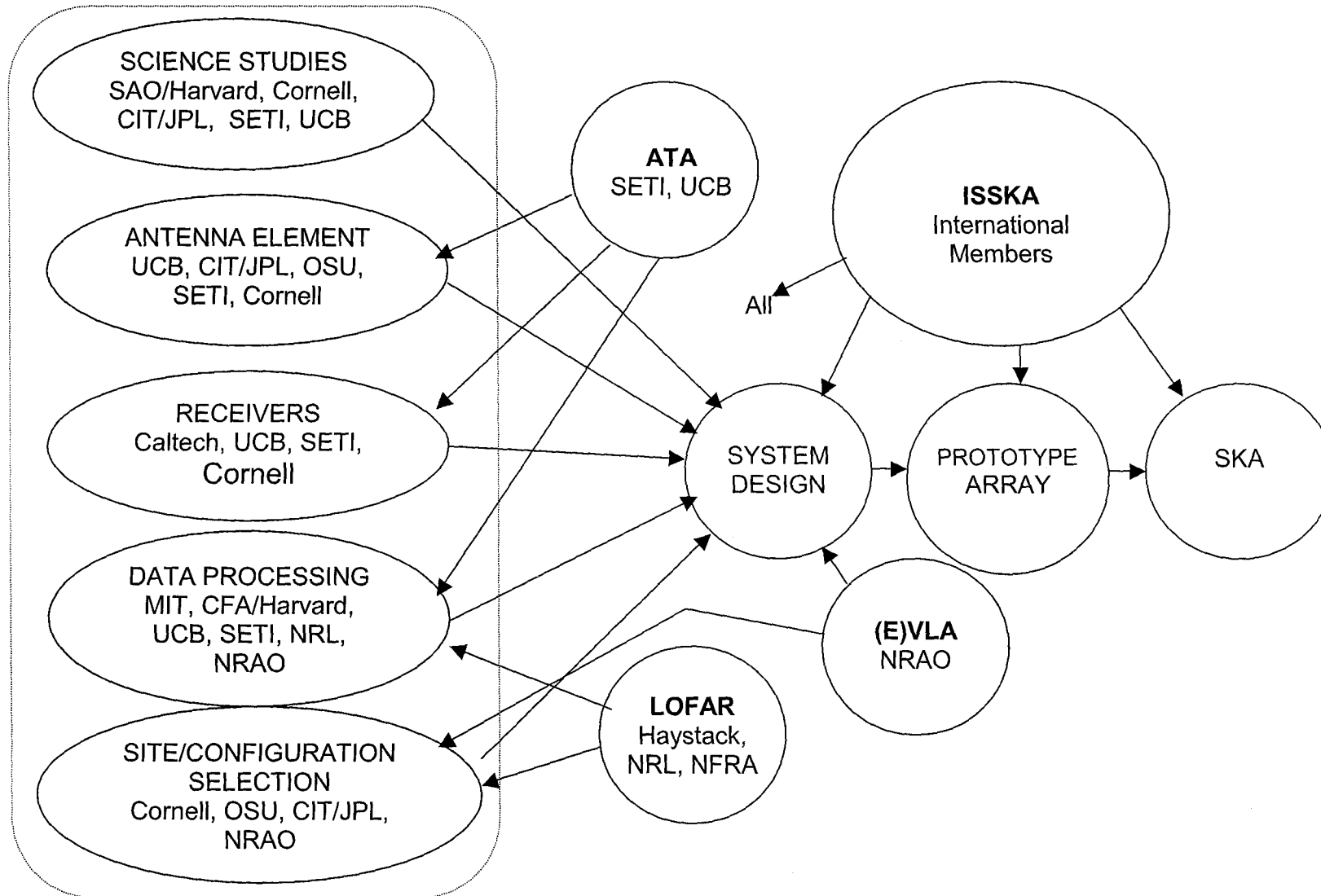
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SKA Technology Summary



USSKA

SKA Organizational Summary

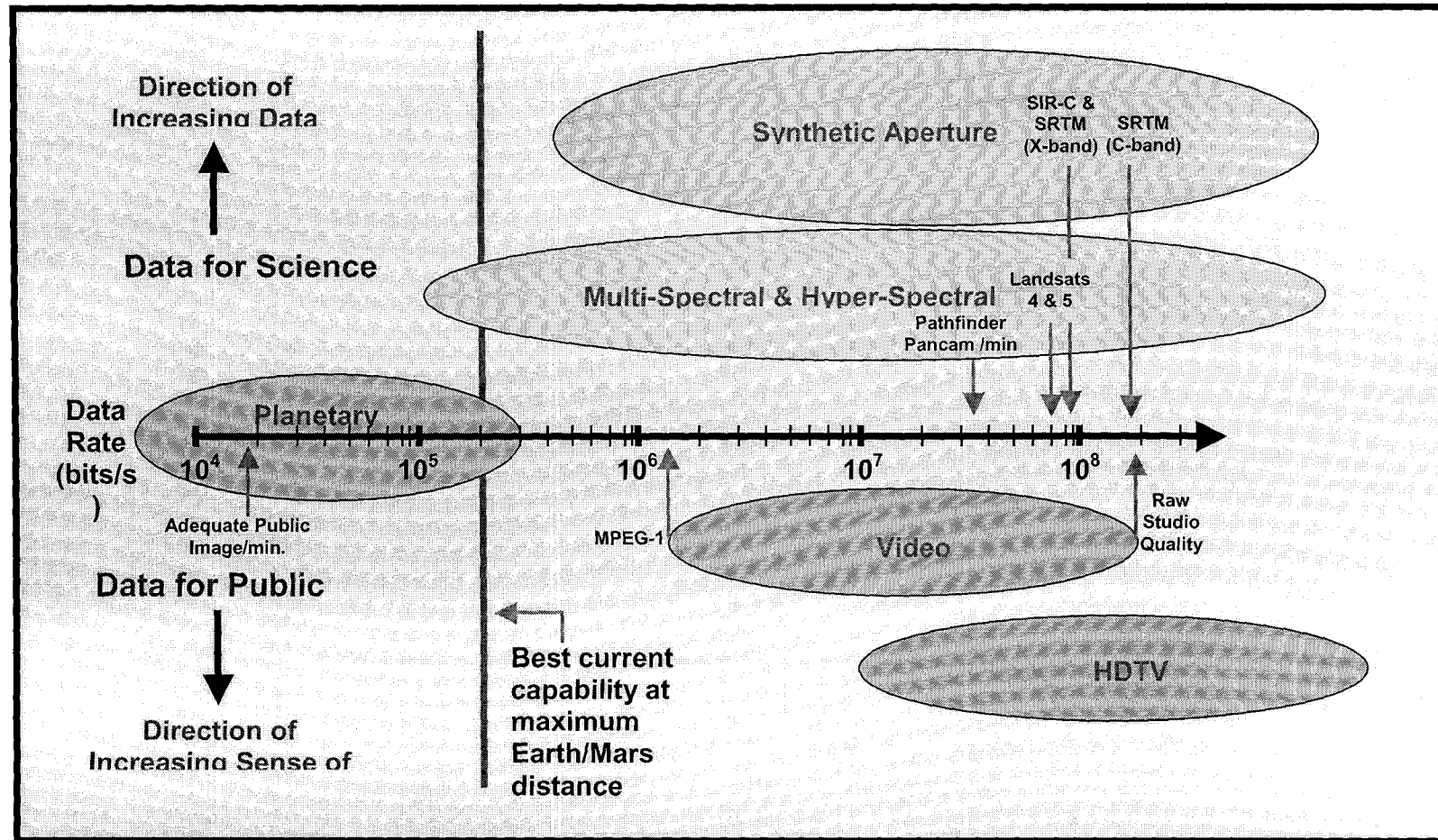


EQUIVALENT IMAGING DATA RATE CAPABILITY AT JUPITER DISTANCE - 750 MILLION KILOMETERS



Increasing Data Rate to Deep Space

- Increasing data rate to deep space will remain the focus for major investment over the next decade
 - It will also be critical to invest in information systems architectures and applications to make use of the increased capability with minimal cost impact to science missions
- Data rate requirements for science and the public are factors of 10 to 100 higher than can be provided with the present DSN



New DSN Technology Thrusts

The future DSN will utilize an optimum combination of new technologies to increase data return, reduce costs, and improve mission robustness

Internet Network Architecture

- Decrease mission costs by adoption of standardized protocols
- Enhance productivity by making mission data easily accessible.
- Enhance reliability by multiple-paths.
- Reuse of software and hardware.

Orbiting Relay Satellites

- Provides orbiting relay with high-gain antenna to increase data transmission rate for outer planet missions
- Allows lander assets to be devoted to science instruments rather than direct to earth communications.

Optical Communications

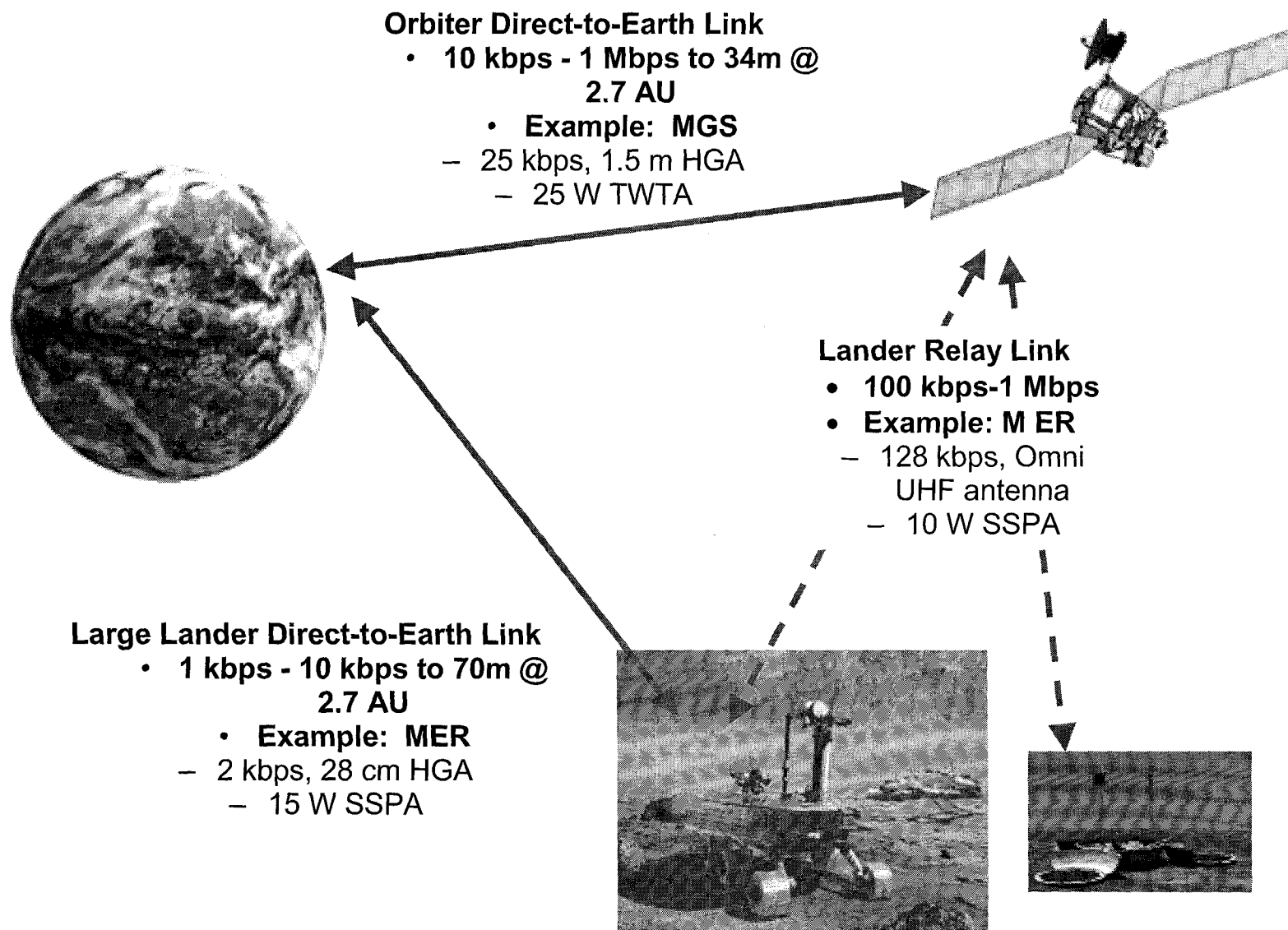
- Very large increase in transmitter gain facilitates very high data rates with low power
- Explosion of new optical communication components and new era of astronomical instruments provides the core technology for optical space communications.
- Challenges are precision pointing, solar radiation, and weather diversity.

Microwave Arrays

- Can provide large increase in data rate by increased receiving aperture on earth
- Minimum cost per unit area by choice of optimum diameter antenna element
- Advantages of multiple beams, soft failure, weather diversity, and expandability
- Synergistic with new developments in radio astronomy such as ATA, SKA, and ALMA.



Mars Telecommunications Examples





SKA Applied to Deep Space Communications



Use SKA to increase data rate by 100+

- Movies, instead of images
- Robotic, virtual-presence exploration of outer planets
- High-resolution multi-spectral imaging
- High-resolution synthetic-aperture imaging of planets
- Short-life, high-data-rate missions to hostile environments (Venus)
- Enables new science missions requiring high data rates

Use SKA to reduce spacecraft mass, power, and cost

- Enables much smaller, lower cost spacecraft
- Reduce or eliminate spacecraft pointing requirements
- Provide low-data monitoring during critical reentry times
- Use to support spacecraft in distress



SKA Applied to Deep Space Navigation



Use SKA to obtain real-time, plane-of-sky, spacecraft position simultaneously with range and Doppler-determined velocity

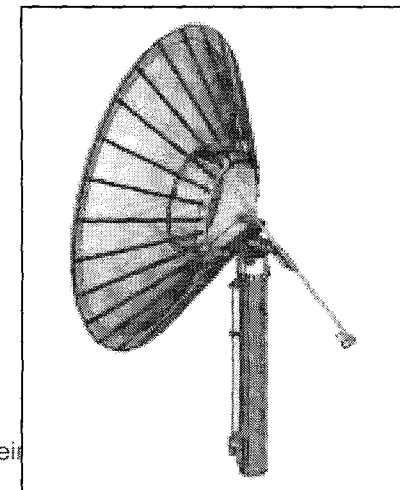
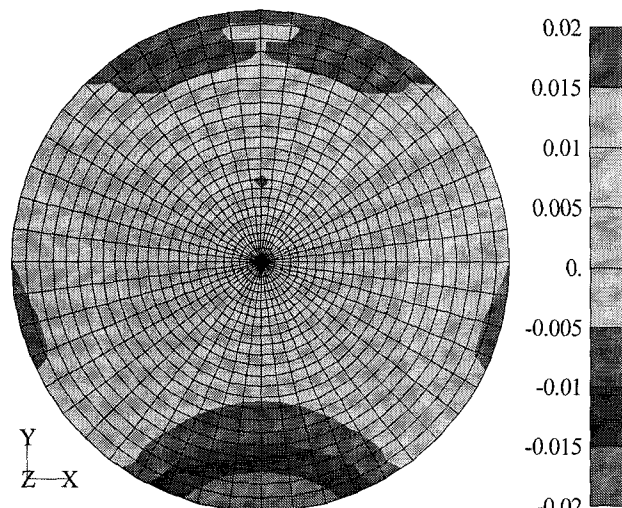
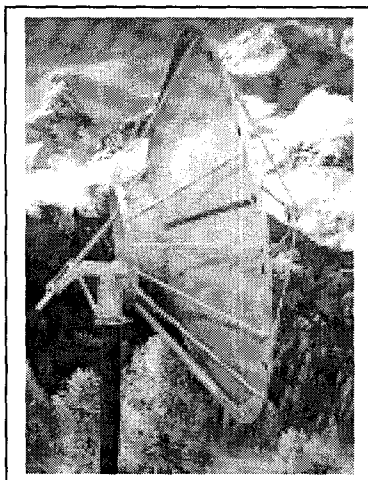
- At 2 AU an array with 500 km baselines and a frequency of 32 GHz has a linear beamwidth of 6km allowing 600m position accuracy.
- The array provides of the order of 5,000 times better angular resolution than the DSN 70m antenna.
- At present spacecraft navigation relies on range and doppler with only coarse angular data. This can result in errors in position determination at critical re-entry or boost orbits as occurred on a recent Mars mission.
- Present VLBI measurements of spacecraft position are not real-time which limits applications.
- An array in multi-beam, multi-frequency transmit mode allows establishment of a distant position reference system without time of flight delay.

Hydroformed Aluminum Antennas

Hydroforming is a process of using a fluid or gas at very high pressure to force aluminum sheet to conform to a mold. The result is a stiff, accurate, and low cost reflector.

JPL has performed a structural analysis of 5m and 8m hydroformed reflectors manufactured by www.anderseninc.com and has found that the wind and gravitational distortions would allow operation at frequencies as high as 100 GHz.

Example	Antenna Diameter	Cost per Antenna	Cost per m ²	Cost per km ²
New 70m DSN antenna	70m	\$100M	\$40.8K	\$40.8B
25m VLBA antenna	25m	\$3M	\$9.6K	\$9.6B
6m ATA antenna	6m	\$30K	\$1.7K	\$1.7B
Target SKA cost	10m	\$30K	\$600	\$0.6B
Hydroformed DBSTV antenna	4m	\$2.8K	\$350	\$0.35B
Aluminum, 3mm thick sheet	Any	NA	\$30	\$.03B

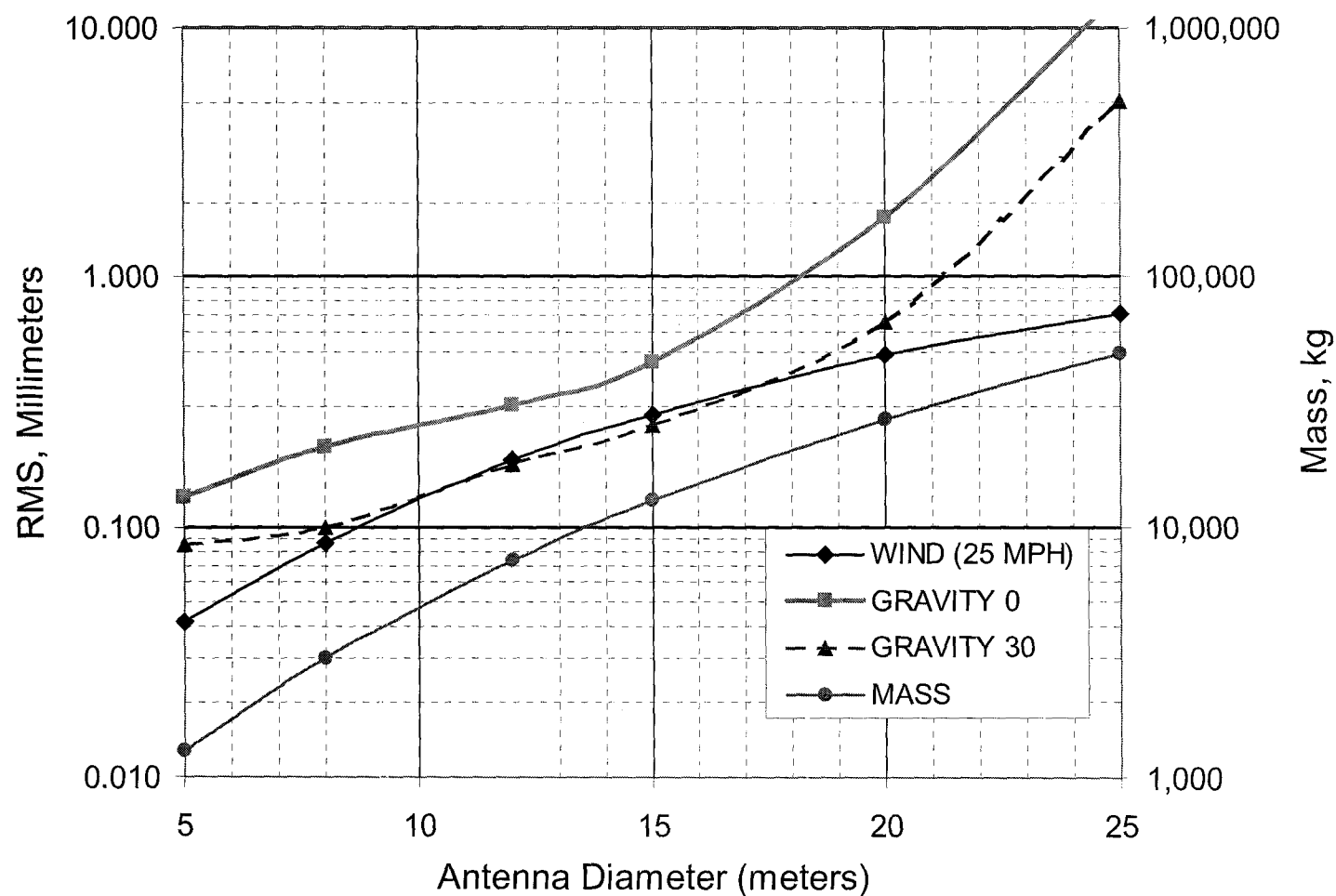


S. Weir

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JPL/Swales Finite-Element CAD Analysis of Hydroformed Shells (0.1mm RMS is Required for an Efficient 100 GHz Antenna)

RMS Deformation Due to Wind and Gravity as a Function of Antenna
Diameter for Hydroformed Shell of 3mm Thickness

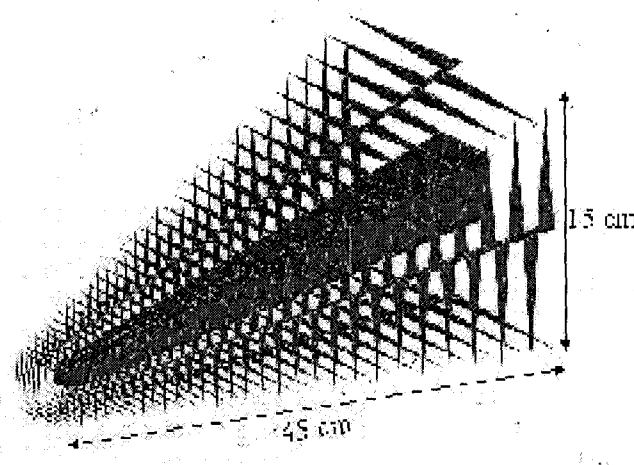


0.5 to 11 GHz Dual-Polarized Feed Developed by SETI/UCB for the ATA

Efficiency > 60% expected over entire frequency range

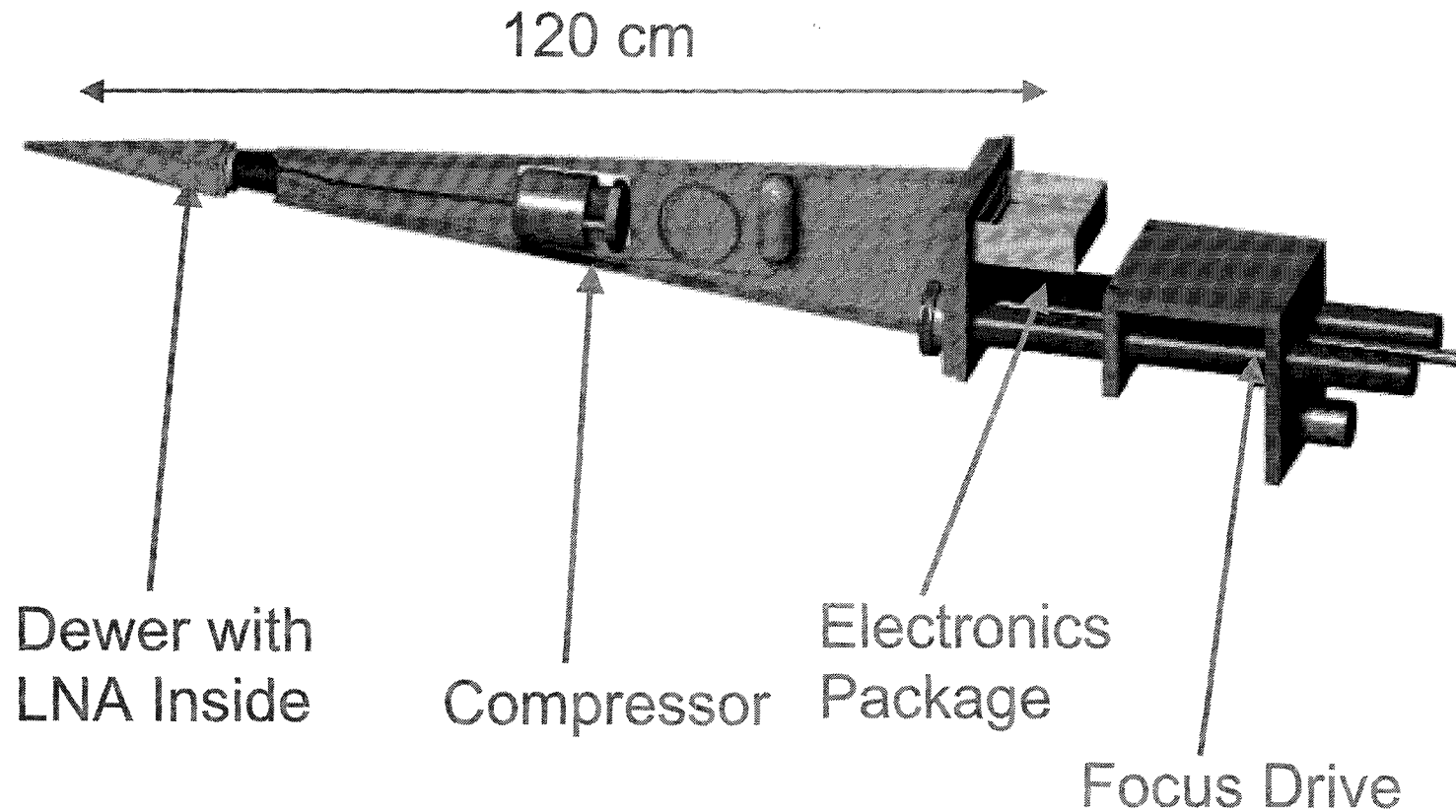


Zig Zag
Log Periodic
Feed



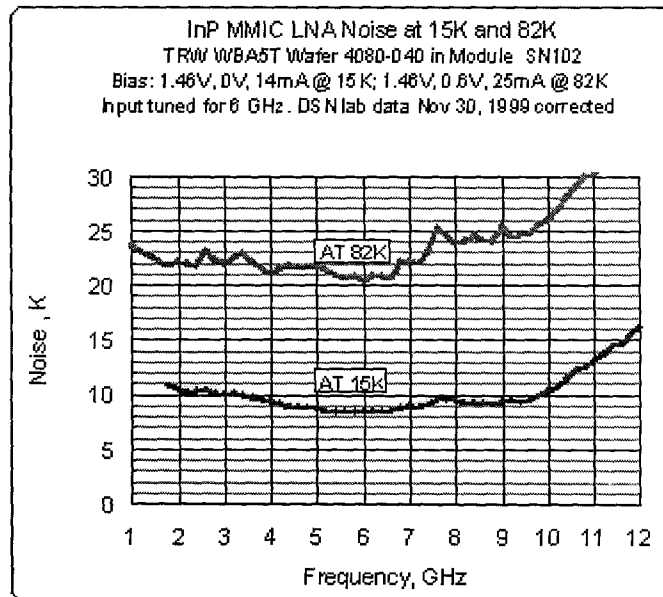
Revolutionary Ultra-Wideband Low Noise Receiver System for the ATA

The structure shown below fits within the pyramidal log-periodic feed and connects to the feed tip. A miniature pulse-tube cryocooler is used to cool the LNA to 60K for <15K noise from 0.5 to 11 GHz.

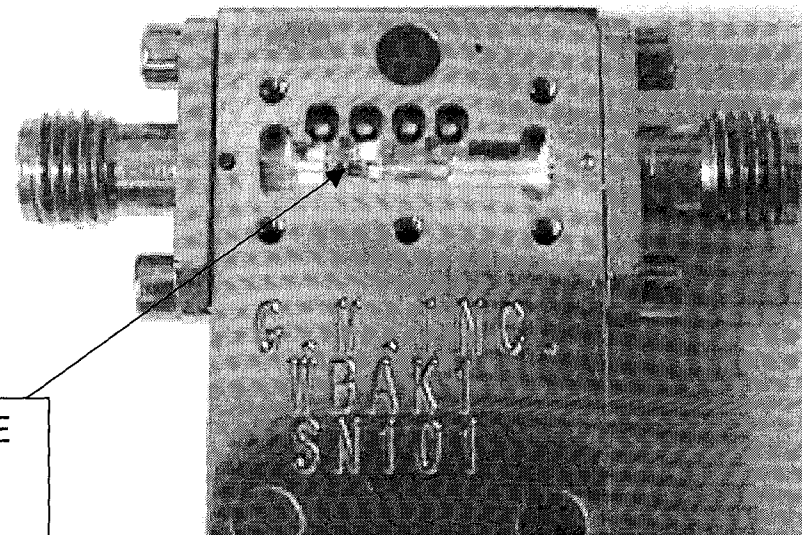


Low-Noise Amplifiers Under Development at Caltech and JPL

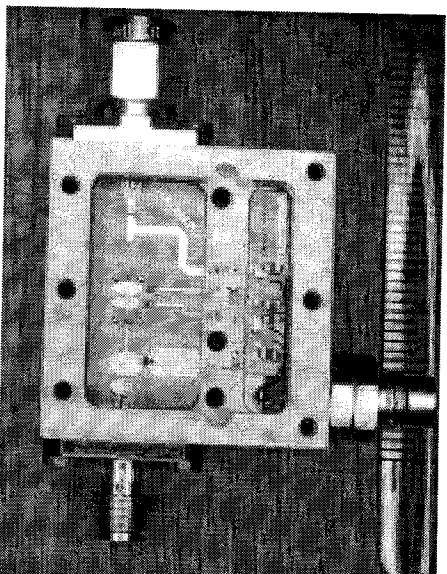
Frequency Range, GHz	Application	Noise
.5-11	ATA	23K @ 80K now, 15K later
4-12	ALMA IF	4K @ 4K, good input match
8-20	SIS IF Amplifier	10K @ 4K, good input match
1-60	NASA Atmospheric Sensor	400K @ 300K, 40K @ 15K
90-110	Planck, Cosmic Background	35K @ 15K
100-140	Atmospheric Sensor	600K @ 300K
170-210	Atmospheric Sensor	1500K @ 300K



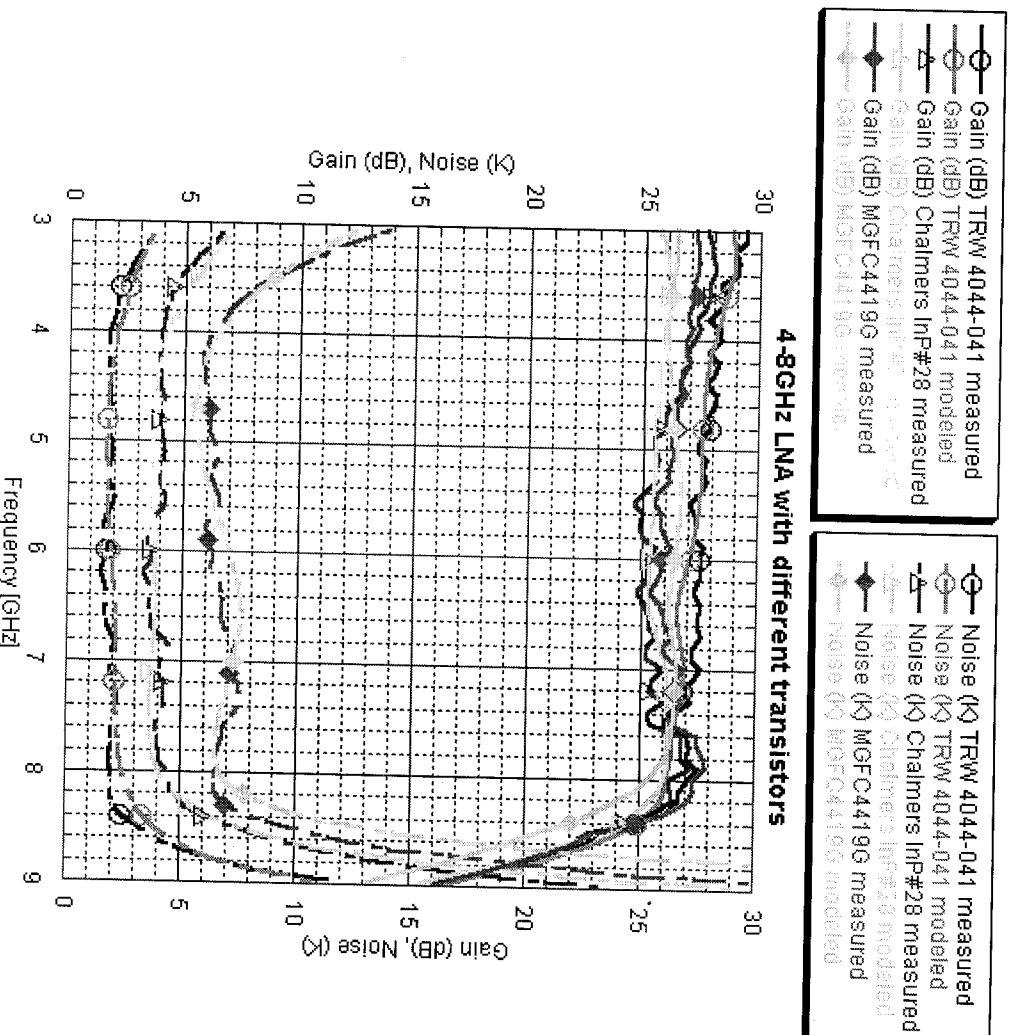
SINGLE
CHIP
LNA



Chalmers 4-8 GHz Cryogenic Low Noise Amplifier



World record 2K noise temperature, measured in 4 laboratories,
achieved with TRW 0.1um InP HEMT



SKA Cost Equation Spread Sheet

Rows are for input or computed parameters such as cost coefficients or performance parameters

Columns are for varying antenna diameters

Sheets are for varying major parameters such as cryogenic temperature or era of electronics costs.

Output of the program is the total cost and cost of each subsystem. The A_{eff}/T_{sys} is constrained at 20,000 by computing the number of antennas needed for each station.

Parameters – There are 14 performance parameters and 19 cost parameters listed on the next page

Models – Details are in the equations in the spreadsheet. Antenna cost is $0.1 \cdot D^3$ K\$.

B&A Cash Estimates July 7, 2007		The (Globe)	
July 2007 (B&A) Cash		The (Globe)	
Category	Amount	Amount	Amount
1.0000	1.0000	1.0000	1.0000
2.0000	2.0000	2.0000	2.0000
3.0000	3.0000	3.0000	3.0000
4.0000	4.0000	4.0000	4.0000
5.0000	5.0000	5.0000	5.0000
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98.0000	98.0000	98.0000	98.0000
99.0000	99.0000	99.0000	99.0000
100.0000	100.0000	100.0000	100.0000

SKA Cost Equation Input Parameters

Units: K\$US(2001), meters, GHz

Param	Array Performance Parameters	Default Value
Ns	Number of stations in array	100
M	Specified Figure of Merit, $M = A/T_{sys}$	20,000
B	Processed total continuum bandwidth	4.0
NI	Number of spectral line channels	16,000
D	Physical diameter of element (meters)	10.0
Ef	Aperature efficiency	0.70
Tant	Antenna noise temperature, $T_{ant} = 10$	14
Kln	Lna noise coefficient dependant upon	0.40
Tphy	Physical temperature of LNA	15
F	Frequency for system temperature spe	10
Nbn	Number of frequency bands	3
Kch	Number of separately digitized chann	4
Le	Average distance, element to station	0.50
Nbeam	Beams per station	4

Param	Array Cost Parameters	Default Value
Cso	Fixed cost per station, land, civil,	300
Ka	Antenna cost coefficient,	0.10
X	Antenna cost exponent	3.0
Ccl	Cooling cost per antenna	20.0
Cfd	Average dual-polariz feed cost	2.0
Cln	Average LNA + mixer cost	0.8
Clo	LO cost	3.0
Cifo	Fixed IF cost per polarization	1.0
Kif	Dual IF cost per GHz of bandwidth	0.2
a1	Digitization coefficient	2.00
e	Digitization exponent	2.00
a2	Digitization constant	0.50
d	Tracking coefficient (per GHz)	0.72
f	Tracking constant	0.10
Kproc	Processing cost coefficient (per	0.48
Kmem	Memory cost (per word)	2.10E-04
kct	Corner turner cost coefficient (p	1.00E-03
Kcor	Correlator cost coef (per baselin	0.024
Kchip	Price per large FPGA chip	0.14

Color code:

Purple: correction of error since previous version.

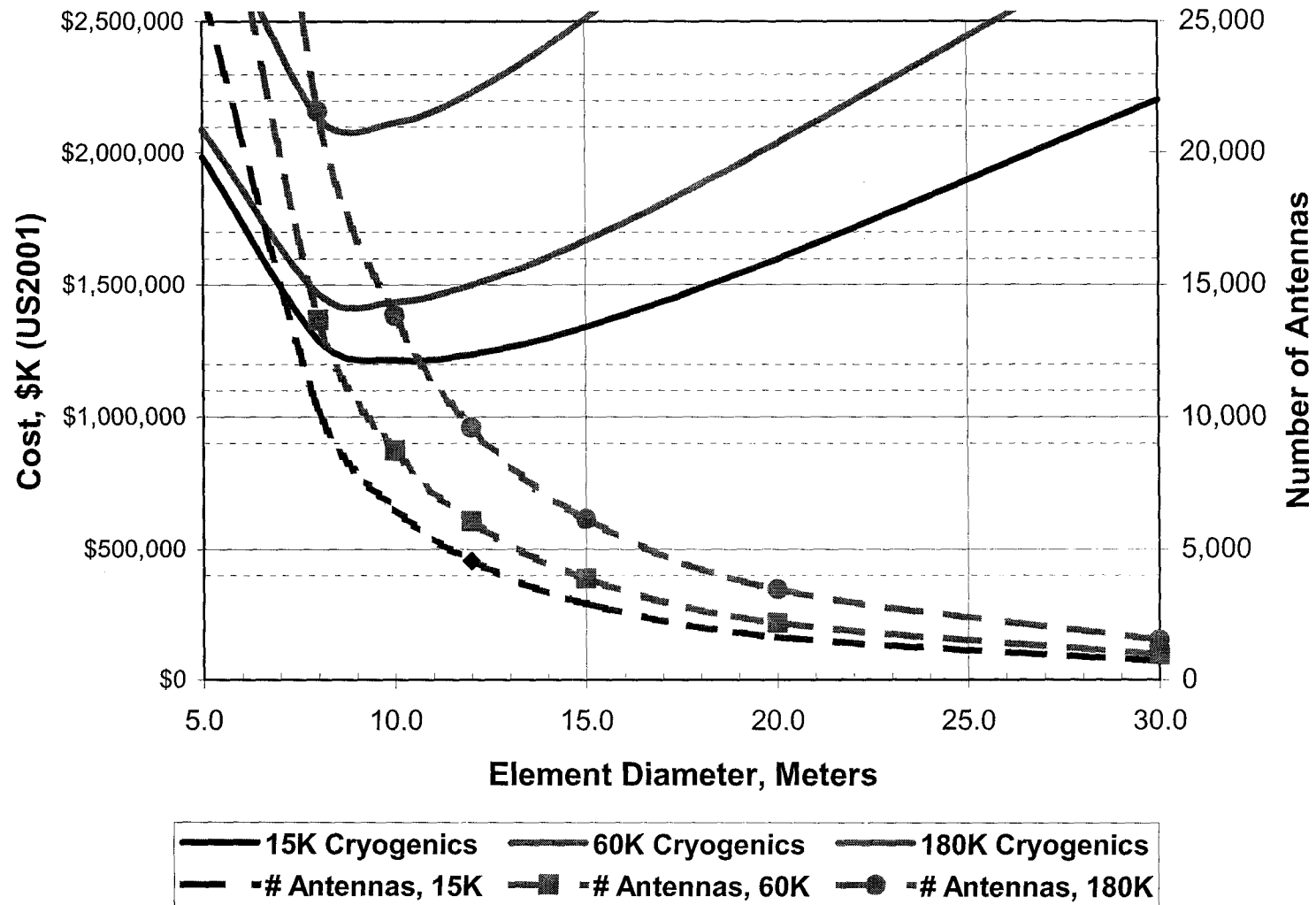
Blue: change to value or formula since previous version.

Red: variable whose value changes across this sheet

Green: variable whose value is different from previous sheet

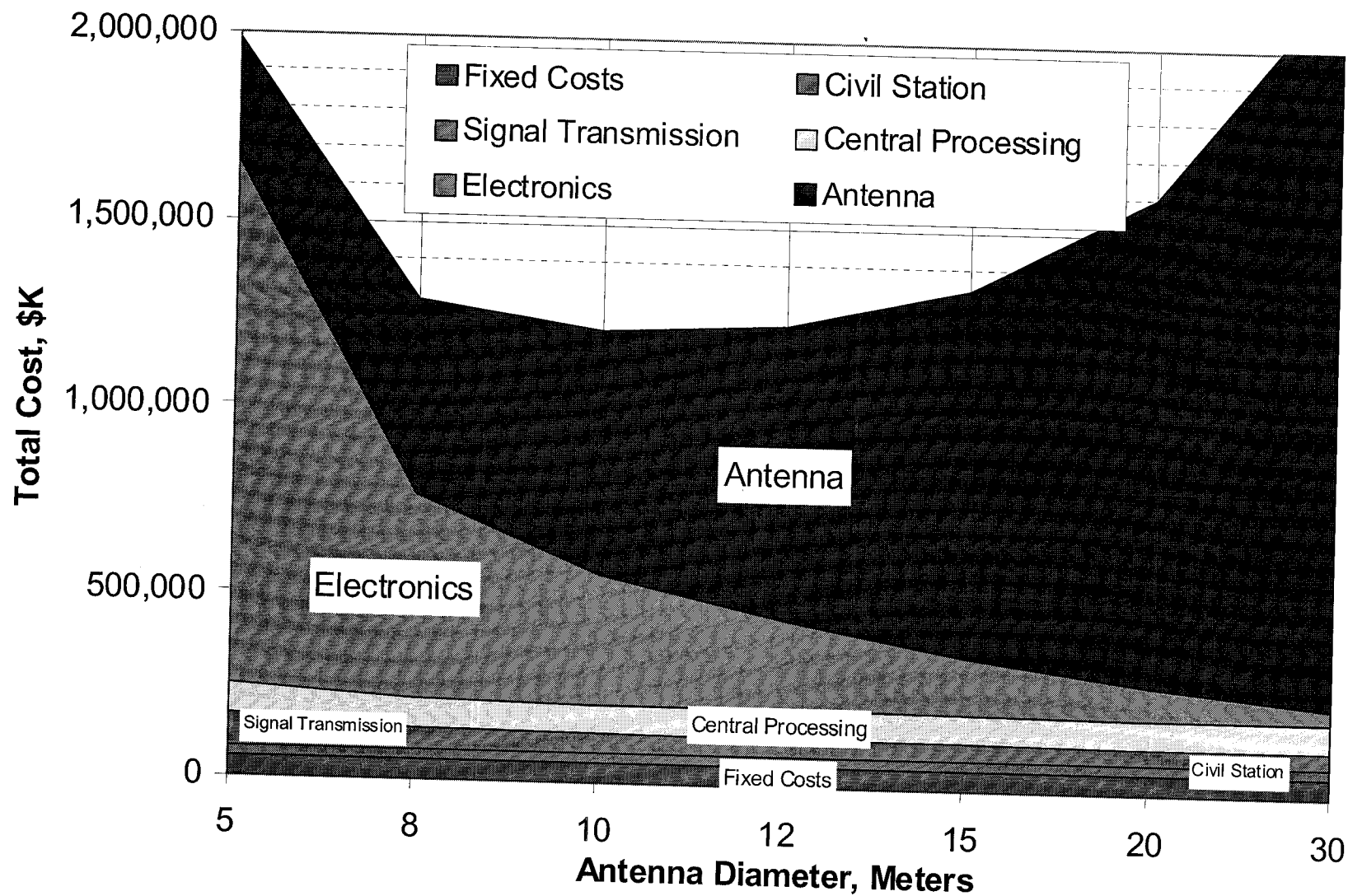
Bold type: input parameter; normal type means value is calculated.

SKA Cost vs Antenna Diameter for 3 Cooling Temperatures
 $A_{\text{eff}}/T_{\text{sys}} = 20,000$, $A_{\text{eff}}=360,000$, $T_{\text{sys}}=18\text{K}$, $\text{BW}=4\text{GHz}$,
 Antenna Cost = $0.1D^3$ K\$, 2001 Electronics Cost = \$54K per Element



SKA Cost Breakdown by Subsystem vs Antenna Diameter

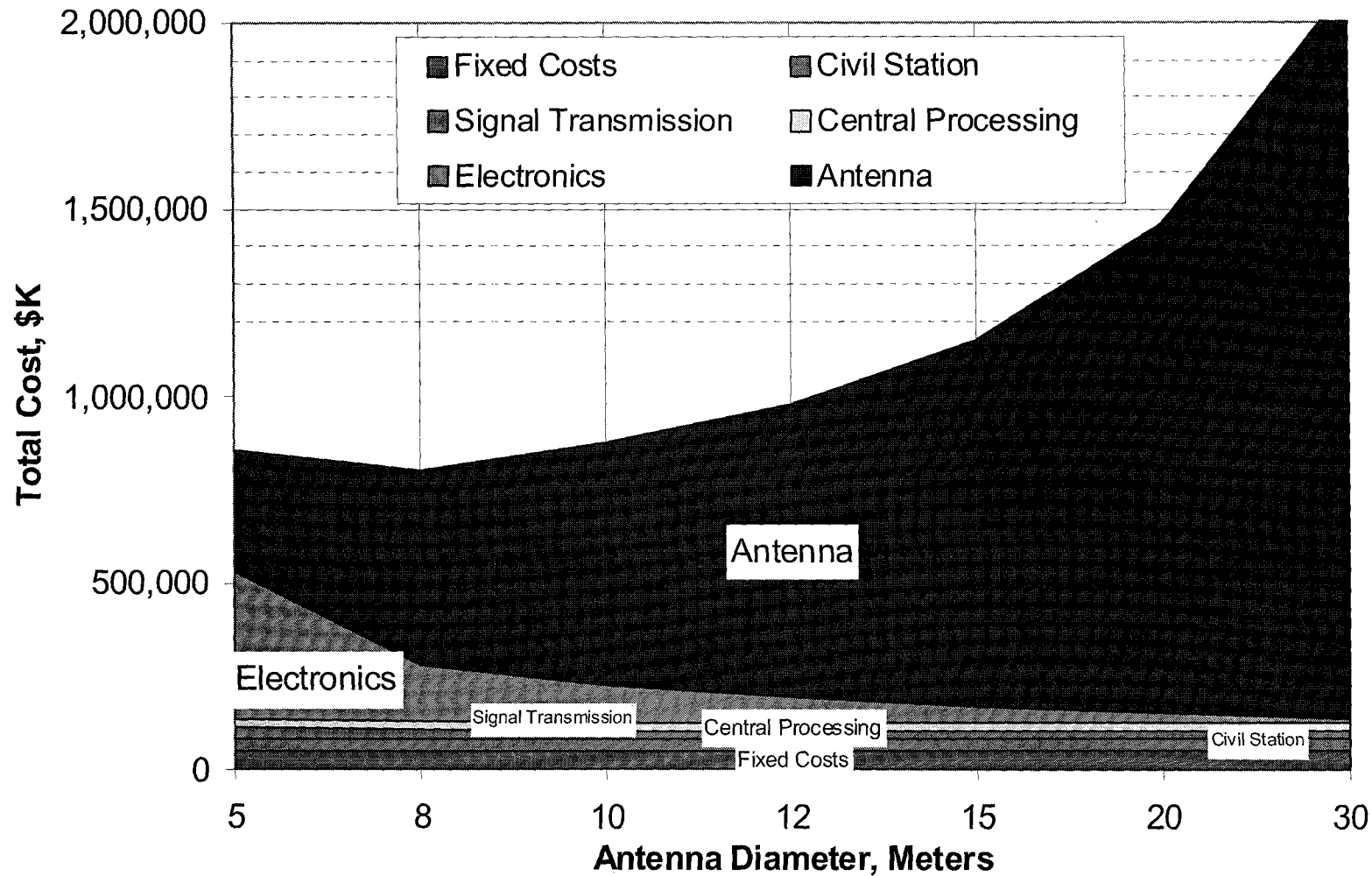
$A_{\text{eff}}/T_{\text{sys}} = 20,000$, $A_{\text{eff}}=360,000$, $T_{\text{sys}}=18\text{K}$, $\text{BW}=4\text{GHz}$, 15K Cryogenics
 Antenna Cost = $0.1D^3$ K\$, 2001 Electronics Cost = \$54K per Element



SKA Cost Breakdown by Subsystem vs Antenna Diameter

$A_{\text{eff}}/T_{\text{sys}} = 20,000$, $A_{\text{eff}}=360,000$, $T_{\text{sys}}=18\text{K}$, $\text{BW}=4\text{GHz}$, 15K Cryogenics

Antenna Cost = $0.1D^3$ K\$, 2010 Electronics Cost = \$15K per Element



SKA for NASA – Overview Summary

- The present DSN capability places severe constraints on deep space science due to data rate limits and spacecraft antenna and power requirements.
- The DSN is saddled with aging facilities which are costly to maintain and is faced with increasing demands from more complex and data-rich missions.
- Construction of antennas larger than 70 or 100m is not cost effective compared with arrays of smaller antennas.
- Compared to the radio astronomy community the DSN has little experience with arrays – a learning period is required to provide know-how and management confidence.
- Long-range development of the DSN is concentrated in optical technology which has limitations due to pointing requirements and high quantum-limited noise.
- There is a ground swell of interest by NASA and JPL scientists in the far-reaching benefits of a SKA to space science. The factor of 100 increase in sensitivity has exciting potential.
- No new space exploration program within NASA wants to put the program at risk by requiring a \$1B SKA yet the collective benefits to many programs over a 30+ year lifetime make very sound business sense.

DSN Situation Summary

- Present data rate from a Mars Lander to a 70m on earth is in the slow modem range – 1 to 10 kbps
- The data rate can be expanded to 10 - 1000 kbps with a Mars orbiting relay.
- The most cost effective way to increase the data rate to either a lander or an orbiting relay is by increasing the collecting area on earth
- Antennas larger than 70 to 100m are not cost effective because cost increases as $D^{2.7}$. A 140m antenna cost more than two 100m antennas.
- Arrays are the best way to increase antenna area in terms of cost, flexibility, and soft-failure. The minimum cost array is one in which the antenna cost is approximately twice the electronics cost. A rough estimate follows:

Electronics Cost per Element	Optimum Antenna	Total Cost For 10 x 70m
\$10M	34m	40 x 34m, \$1.0B
\$1M	17m	160 x 17m, \$0.5B
\$100K	8.5m	640x8.5m, \$0.3B

Objectives of a JPL DSN Array Technology Development Program

- **Liaison** with SKA, ATA, and eVLA array development tasks; assess technology and joint use for NASA applications.
- **Determine array cost vs antenna diameter** and electronics cost.
- **Verify cost** by fabrication and test of an array element followed by a prototype array.
- **Perform up-link phasing** experiments to gain experience and assess calibration burden.
- **Develop management confidence** in communication arrays by forming a group with the core expertise needed for implementation.

Goals for a JPL SKA/DSN Development Program

System Design

- Design interfaces to DSN user and science community
- Find optimum antenna element size and array configuration
- Participate in the SKA development

Antenna Manufacture

- Develop manufacturing technology to reach \$600/m² cost target
- Rotating antenna turntable – conceptual design and cost estimate.
- Monitor ATA 350x6m manufacture
- Construct small antenna prototypes to verify cost and performance

Transmitter Design

- Design, fabricate, and test solid state 7 and 33 GHz HPA's (100W, 5W)
- Develop diplexer and exciter
- Develop geosynchronous satellite monitor receiver for transmitter phasing

Receiver Development

- Develop low cost MMIC LNA's , feed, and cryogenics; target \$10K

Connectivity

- Develop satellite relay LO distribution and fiber-optic signal transmission

Signal Processing

- Design and prototype digital beam former and correlator

21st Century DSN Visions

The DSN should not only support, but should stimulate and enable the exploration of the solar system with large increases in data flow to both scientists and the public.

Year	Developmental Milestones	Operational System	Landmark Capability
2010	Optical link demos Prototype array demo	Robust 70m equivalents at Ka band at 3 DSN sites. IPN protocols in use	Reliable service to all planned missions in decade. Routine use of orbiting relays.
2020	Robust, low-cost 8 Gbt/s, W-band transceivers	10 x 70m equivalent array performance; optical links to outer planets	Virtual presence HDTV exploration of outer planets
2030	Interstellar transmitters, MEMs based receivers, Single-electron transistors	Square-KM arrays and optical links at 3 sites;	Detection of radio activity from identified earth-like planets